On The Model Reduction for Chemical and Physical Kinetics

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PhD Thesis Presentation 18th August 2015 Introduction

2 gRRM

3 Entropy Prod. Ana.

4 n-heptane/air complex dynamics



Outline

Introduction and Motivation

The global Relaxation Redistribution Method

Entropy production analysis for mechanism reduction

n-heptane/air complex dynamics

Summary and future works

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Challenges and Motivation

Every Mesh Grid, Every Time Step

- 1. Mass Conservation Equation
- 2. Momentum Conservation Equations
- 3. Energy Conservation Equation
- 4. ns PDEs for temporal evolution of ns species

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Challenges and Motivation

► Every Mesh Grid, Every Time Step

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Size of detailed chemical kinetics





CH₄ C7H16 C10H22 C12H26 C20H42-2 561 1282 7200 940 325 2539 3878 5030 31400 Sizes of detailed reaction mechanisms for sample hydrocarbons

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Challenges and Motivation

► Every Mesh Grid, Every Time Step

- 1. Mass Conservation Equation
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Stiffness/ Non-linearity



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Computational Cost Reduction for Chemical Kinetics

A Time scale analysis:

Describe chemistry using fewer variables.

- CSP: Lam & Goussis (1989)
- ILDM: Maas & Pope (1992)
- ✓ MIM: Karlin & Gorban (1991)

RRM: Kooshkbaghi et al. (2014)



Chem. Kin. Model Reduction

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Computational Cost Reduction for Chemical Kinetics

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B Conventional Reduction Methodology:

Generate smaller skeletal mechanisms from the detailed mechanism by systematically removing unimportant species and reactions.

- ✓ CSP: Massias et al. (1999)
- ✓ DRG: Lu & Law (2005)
- ✓ PFA: Sun et al. (2010)

Entropy Production Analysis: Kooshkbaghi et al. (2010)

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Computational Cost Reduction for Chemical Kinetics

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Entropy Production Analysis: Kooshkbaghi et al. (2010)

C Storage and Retrieval methods

- (ISAT: Pope (1997)
- PRISM: Tonse (2003)

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Thesis Outline





Skeletal mechanism generation and application



Ch. 5* Complex dynamics of heavy hydrocarbon * Bifurcation analysis * Reactions supporting and opposing critical behaviour

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large fuels

* Bifurcation analysis

critical behaviour

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Thesis Outline



Thesis in a nutshell

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5 Conclusion

II. The global Relaxation Redistribution Method



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- ► In dynamical system $\{\mathbb{T}, S, \phi^t\}, U \subset S \text{ is invariant}$ manifold (set) if $N_0 \in U$ then $\forall t: \phi^t \mathbf{N}_0 \in U$
- Slow Invariant Manifold (SIM)

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Concept of Slow Invariant Manifold

Classification of Systems

- Autonomous system
- Cauchy-Lipschitz

 $\frac{d\mathbf{N}}{dt} = f(\mathbf{N})$ $f : \mathbb{R}^{n_s} \supset S \to \mathbb{R}^{n_s}$ $\mathbf{N} \in S, t \in \mathbb{T}$ $\phi^t_{t \in \mathbb{T}} \text{ is called a flow where}$

 $\mathbf{N}_t = \boldsymbol{\phi}^t \mathbf{N}_0$

 \mathbf{N}^{eq} is a unique fixed point.

In dynamical system {T, S, φ^t}, U ⊂ S is invariant manifold (set) if N₀ ∈ U then ∀t: φ^tN₀ ∈ U

Slow Invariant Manifold (SIM)



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Multiscale Dissipation

Davis-Skodje System (J. Chem. Phys. 1999):

$$\begin{cases} \frac{dx}{dt} = -x\\ \frac{dy}{dt} = -\gamma y + \frac{(\gamma - 1)x + \gamma x^2}{(1 + x)^2}\\ \gamma \gg 1, \gamma = 5 \end{cases}$$



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Multiscale Dissipation

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$$\begin{cases} \frac{dx}{dt} = -x\\ \frac{dy}{dt} = -\gamma y + \frac{(\gamma - 1)x + \gamma x^2}{(1 + x)^2}\\ \gamma \gg 1, \gamma = 5 \end{cases}$$

$$x = x_0 e^{-t}$$

$$y = \frac{x_0 e^{-t}}{1 + x_0 e^{-t}} + \frac{C(x_0, y_0) e^{-y_0 t}}{y_{slow}} = \frac{x}{1 + x}$$

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Multiscale Dissipation

Davis-Skodje System (J. Chem. Phys. 1999):

$$\begin{cases} \frac{dx}{dt} = -x\\ \frac{dy}{dt} = -\gamma y + \frac{(\gamma - 1)x + \gamma x^2}{(1 + x)^2}\\ \gamma \gg 1\\ \implies y_{SIM} = \frac{x}{1 + x} \end{cases}$$



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Method of Invariant Manifold (MIM)

$$f(\mathbf{N}(\xi)) = f(\mathbf{N}(\xi))_{\parallel_{T_W}} + f(\mathbf{N}(\xi))_{\perp_{T_W}}$$

$$f(\mathbf{N}(\xi))_{\parallel_{T_W}} = \mathbf{P}f(\mathbf{N}(\xi))$$

$$f(\mathbf{N}(\xi))_{\perp_{T_W}} = \Delta = (\mathbf{I} - \mathbf{P})f(\mathbf{N}(\xi))$$
Invariance Condition
$$\Delta = 0, \qquad \xi \in \Xi$$
MIM: The slow invariant manifold
is the stable solution of the film
extension of dynamics:
$$\frac{d\mathbf{N}(\xi)}{dt} = \Delta$$

Gorban, A. N., & Karlin, I. V. (2004). Lect. Notes Phys., 660.

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Relaxation Redistribution Method (RRM)

1. Find/Choose the slow parameterization variables



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Relaxation Redistribution Method (RRM)

- 1. Find/Choose the slow parameterization variables
- 2. Construct the initial guess of slow manifold



RhisshRoghi, Nr. Kaslin (2014) 29.1 Chemlyph Bs. 141 (43) (344) 362.06.

2 gKKIM SIM Def. MIM RRM Algorithm

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Relaxation Redistribution Method (RRM)

- 1. Find/Choose the slow parameterization variables
- 2. Construct the initial guess of slow manifold
- 3. Relax all the points on the initial manifold



Rbisstazzagli; Nr. Kerlin (2014) 29.1 Chenlyph B. v. 141 (43) 644962.06.

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> RRM Algorithm Constructing Manifold Using Manifold

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Relaxation Redistribution Method (RRM)

- 1. Find/Choose the slow parameterization variables
- 2. Construct the initial guess of slow manifold
- 3. Relax all the points on the initial manifold
- 4. Points moving toward local equilibrium manifold



Kbiastakagafi; N. Kaalin (2014) 29.1 Cherky phys. 141 (43) 644962.06.

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Relaxation Redistribution Method (RRM)

- 1. Find/Choose the slow parameterization variables
- 2. Construct the initial guess of slow manifold
- 3. Relax all the points on the initial manifold
- 4. Points moving toward local equilibrium manifold
- 5. Redistribute back to neutralize slow motion
 - Redistribution : Interpolation for interior
 - Redistribution : Extrapolation for missing point



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Kbiastakagelfi; Mr. Kastlin (2014) 29.1 Chernhyph Bey. 141 (43) 644962.06.

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RRM Manifold Construction

Singular perturbed system*

$$\begin{array}{l}
\frac{dx}{dt} = 2 - x - y \\
\frac{dy}{dt} = \gamma(\sqrt{x} - y) \\
\gamma \gg 1
\end{array}$$

Relaxation Step



*KESSHRBanin, M. etal., (2014). J. Newen. Phys., 1448(4),8343702.

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RRM Manifold Construction

Singular perturbed system*

$$\frac{dx}{dt} = 2 - x - y$$

$$\frac{dy}{dt} = \gamma(\sqrt{x} - y)$$

$$\gamma \gg 1$$

Redistribution Step



A B > 4
 B > 4
 B

4 ∃ ≥

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RRM Manifold Construction

Singular perturbed system*

$$\begin{cases} \frac{dx}{dt} = 2 - x - y\\ \frac{dy}{dt} = \gamma(\sqrt{x} - y)\\ \gamma \gg 1 \end{cases}$$

▶ ILDM manifold is neither invariant nor slow for $0 \le x \le 0.7$.



*KESSHRBanin, M. etal., (2014). J. Newen. Phys., 1448(4),8343702.

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Dimensionality issues

Computational cost

- Manifolds are represented on a grid
- Retrieving data of high dimensional tables, imposes restrictions on the dimension
- \implies Target : 2D/3D manifold

Pope, S. B. (2013). Proc. Combust. Inst., 34(1), 1-31.

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Dimensionality issues

Computational cost

- Manifolds are represented on a grid
- Retrieving data of high dimensional tables, imposes restrictions on the dimension
- \implies Target : 2D/3D manifold
- Dimension of SIMs
 - SIMs usually limited to a small neighborhood around equilibrium

 \implies How to extend it further to cover the states all the way to the fresh mixture?

Pope, S. B. (2013). Proc. Combust. Inst., 34(1), 1-31.

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Dimensionality issues

 \implies How to extend it further to cover the states all the way to the fresh mixture?

Construct the Slow invariant manifold and extend via prolongation with linear extrapolation*



*Bykov, V., & Maas, U. (2007). Proc. Combust. Inst., 31(1), 465-472.

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Dimensionality issues

 \implies How to extend it further to cover the states all the way to the fresh mixture?

Construct the Slow invariant manifold and extend via prolongation with linear extrapolation*



Construct the initial grid which covers the admissible solution space and refine it via RRM, (global RRM**) *Bykov, V., & Maas, U. (2007). Proc. Combust. Inst., 31(1), 465-472. **Kooshkbaehi, M. et al. (2014). J. Chem. Phys., 141(4).044102.

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global RRM (gRRM)

- 1. Construct 2D initial grid (Ξ will be defined later)
- 2. Find the boundaries of initial grid



MIM Constructing Dimensional

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global RRM (gRRM)

- 1. Construct 2D initial grid (E will be defined later)
- 2. Find the boundaries of initial grid



- 3. Relax interior points
- 4. Redistribute back points on initial grid via interpolation of scattered grid

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global RRM (gRRM)

The gRRM *n*_d-dimensional manifold is:

- The n_d -dimensional SIM +
- The extension of SIM to the far from equilibrium states

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global RRM (gRRM)

The gRRM n_d -dimensional manifold is:

- The n_d -dimensional SIM +
- The extension of SIM to the far from equilibrium states

 \implies The initial grid is important both for convergence and accuracy of extension

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global RRM (gRRM)

The gRRM n_d -dimensional manifold is:

- The n_d -dimensional SIM +
- > The extension of SIM to the far from equilibrium states
- \implies The initial grid is important both for convergence and accuracy of extension
- \implies In this work the initial grid is found based on notation of Quasi Equilibrium Manifold (QEM)

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Initial Grid via QEM (CEM)

$$\begin{array}{ll} \min & G \\ \text{s.t.} & \mathbf{BN} = \boldsymbol{\xi} \end{array}$$

$$\mathbf{B} = [\mathbf{E} \ \mathbf{B}^d]$$
 and $\boldsymbol{\xi} = [\boldsymbol{\xi}^e \ \boldsymbol{\xi}^d]$

• $n_e \times n_s$ elemental constraints matrix, **E**

 $\mathbf{EN} = \xi^e$

 ξ^{e} is specified by the initial composition E_{ji} : number of atoms of element *j* in species *i*

• $n_d \times n_s$ constraints matrix \mathbf{B}^d

$$(\mathbf{B}^d)\mathbf{N} = \boldsymbol{\xi}^d$$

 ξ^d : slow parameters **B**^d: rows define the linear combination of **N** as the slow constraints 1 Introduction

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Manifold parametrization Ξ / Proper choose for extension

How to Choose a good set of constraints \mathbf{B}^d ?

- Rate-Controlled Constrained-Equilibrium method (RCCE)
- constraints for H₂/air combustion with n_s = 9 species and n_r = 21 elementary reactions*

| ** | H_2 | N_2 | Н | 0 | OH | O_2 | H_2O | HO_2 | H_2O_2 |
|-------------------------|-----------|---|--|---|--|---|--|---|--|
| | 1 | 1 | 1 | 1 | 1 | 1 | 1 | 1 | 1 |
| ξ_2 =Active Valence | | 0 | 1 | 2 | 1 | 0 | 0 | 0 | 0 |
| | 0 | 0 | 0 | 1 | 1 | 0 | 1 | 0 | 0 |
| | | | | | | | | | |
| 1 | 1 | 1 | 1 | 1 | 1 | 1 | 1 1 | 1 | |
| 0 | 0 | 1 | 2 | 1 | 0 | 0 | 0 0 | | |
| 0 | 0 | 0 | 1 | 1 | 0 | 1 | 0 0 | | |
| | e [1 0 0 | $ \begin{array}{cccccccccccccccccccccccccccccccccccc$ | $\begin{bmatrix} 1 & 1 & 1 \\ 0 & 0 & 1 \\ 0 & 0 & 1 \\ 0 & 0 & 0 \end{bmatrix}$ | $\begin{bmatrix} 1 & 1 & 1 & 1 \\ 0 & 0 & 1 \\ 0 & 0 & 0 \end{bmatrix}$ | $\begin{bmatrix} 1 & 1 & 1 & 1 & 1 \\ 0 & 0 & 1 & 2 \\ 0 & 0 & 0 & 1 \end{bmatrix}$ $\begin{bmatrix} 1 & 1 & 1 & 1 & 1 \\ 0 & 0 & 1 & 2 & 1 \\ 0 & 0 & 1 & 2 & 1 \\ 0 & 0 & 0 & 1 & 1 \end{bmatrix}$ | $ \begin{array}{cccccccccccccccccccccccccccccccccccc$ | $\begin{bmatrix} 1 & 1 & 1 & 1 & 1 & 1 & 1 \\ 0 & 0 & 1 & 2 & 1 & 0 \\ 0 & 0 & 0 & 1 & 1 & 0 \end{bmatrix}$ $\begin{bmatrix} 1 & 1 & 1 & 1 & 1 & 1 \\ 0 & 0 & 1 & 2 & 1 & 0 \\ 0 & 0 & 1 & 2 & 1 & 0 & 0 \\ 0 & 0 & 0 & 1 & 1 & 0 & 1 \end{bmatrix}$ | $\begin{array}{c ccccccccccccccccccccccccccccccccccc$ | $\begin{bmatrix} 1 & 1 & 1 & 1 & 1 & 1 & 1 & 1 & 1 \\ 0 & 0 & 1 & 2 & 1 & 0 & 0 & 0 \\ 0 & 0 & 0 & 1 & 1 & 0 & 1 & 0 \\ \end{bmatrix}$ |

*kirahgtol; &20042, 41 B. (2004): Kimbilst 9hebby Model., 8(2), 255-279.

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gRRM for H_2/air combustion

 $n_s = 9$ Species, $T_0 = 1500 K$, P = 1 atm, $\phi = 1.0$



ξ_1 =Total Mole, ξ_2 =Active Valence Slight improvements in main species

□, fresh mixture; ★, equilibrium point; – detailed kinetics path, Colored surfaces, Manifolds

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gRRM for H_2/air combustion

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gRRM for H_2/air combustion





 ξ_1 =Total Mole, ξ_2 =Active Valence Large improvements for low-concentration radicals

□, fresh mixture; ★, equilibrium point; – detailed kinetics path, Colored surfaces, Manifolds

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H_2/air auto-ignition

Adiabatic, constant pressure reactor $T_0 = 1500 K, P = 1 atm, \phi = 1.0$ **2D** manifold results



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SIM I

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Constructing Manifold

sing Manifold

 $T_0 = 1500 \text{K}$

 $T_0=1000\mathrm{K}$

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4 n-heptane/air complex dynamics

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H_2/air auto-ignition

Adiabatic, constant pressure reactor $T_0 = 1000 K, P = 1 atm, \phi = 1.0$ **2D** manifold results



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H_2/air auto-ignition

Adiabatic, constant pressure reactor $T_0 = 1000 K, P = 1 atm, \phi = 1.0$ **3D** manifold results



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III. Entropy production analysis for mechanism reduction

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Entropy Production



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Entropy Production

$$\sum_{i=1}^{n_s} v_{ik}' N_i \rightleftharpoons \sum_{i=1}^{n_s} v_{ik}'' N_i, \quad k = 1, \cdots, n_r$$

$$q_k = q_{f_k} - q_{r_k} = k_{f_k} \prod_{i=1}^{n_s} [N_i]^{v'_{ik}} - k_{r_k} \prod_{i=1}^{n_s} [N_i]^{v''_{ik}}, \quad k = 1, \cdots, r$$

The entropy production per unit volume

$$\frac{1}{V}\frac{dS}{dt} = R_c \sum_{k=1}^{n_r} (q_{f_k} - q_{r_k}) ln\left(\frac{q_{f_k}}{q_{r_k}}\right)$$

The relative contribution of each reaction in total entropy production at time *t*

$$r_k(t) = \frac{R_c(q_{f_k} - q_{r_k})ln\left(\frac{q_{f_k}}{q_{r_k}}\right)}{\frac{1}{V}\frac{dS}{dt}}$$

Threshold for contribution

$$r_k(t) \geq \varepsilon\%$$

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Most-Contributing Reactions

n-heptane LLNL2 Mechanism $(n_s = 561, n_r = 2539)^*$ $T_0 = 650$ K, P = 1 atm, $\phi = 1$ $\varepsilon = 5\%$



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Generate Skeletal Mechanism

Most-contributing reactions



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n-heptane/air complex dynamics



Generate Skeletal Mechanism

Most-contributing reactions



Important Species

 nc_7h_{16} , o_2 , c_7h_{15} -1/2/3/4, ho_2 , oh, nc_7ket_{24} , nc_7ket_{35} , nc_7ket_{42} , ...

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Generate Skeletal Mechanism

Most-contributing reactions



Important Species

 $nc_7h_{16}, o_2, c_7h_{15}-1/2/3/4,$ $ho_2, oh, nc_7ket_{24}, nc_7ket_{35},$ nc_7ket_{42}, \cdots

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Skeletal Mechanism Generation

- Eliminate non-important species
- Keep all elementary reactions including important species

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Skeletal Mechanism for *n*-heptane/air kinetics

- Detailed Mechanism (D561): $n_s = 561$, $n_r = 2539$
- Sampled points took from autoignition in adiabatic constant pressure reactor
 - ▶ $650 \le T_0 \le 1400 \text{ K}$
 - $1 \le P \le 20$ atm
 - $0.5 \le \phi \le 1.5$
- $\varepsilon = 0.2\% \longrightarrow n_s = 203, n_r = 879$ (R203)
- $\varepsilon = 0.6\% \longrightarrow n_s = 149, n_r = 669 \text{ (R149)}$

Kooshkbaghi, M., et al., (2014). Combust. Flame, 161(6), 1507-1515.

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Validation : Ignition Delay



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Stiffness



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Validation : Single-Zone Engine Model

 $T_{inlet} = 650 \text{ K}, P_{inlet} = 5 \text{ atm}, \phi = 0.8 \text{ at} -40 \text{ }^{\circ}\text{ATDC}, \omega = 700 \text{ rpm}$



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Validation : Laminar Premixed Flame

 $T_u = 650 \text{ K}, P = 1 \text{ atm}$



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IV. *n*-heptane/air complex dynamics

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PSR Setup Validation of Ske. Mech. 1P Cont. Multi-P Cont.

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 τ : Residence Time p: Reactor Pressure φ: Inlet Mixture T_0 : Inlet Temperature \dot{Q}_{loss} : Heat loss per unit volume

AUTO-07p: Continuation and Bifurcation Software for ODEs + CHEMKIN III: Chemical kinetics data

PSR Setup



Validation Of Skeletal Mechanism



Dependence of reactor temperature on residence time of adiabatic PSR $T_0 = 650 \text{ K}, \phi = 1.0$

D561 (solid lines) and R149 (open circles)

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One parameter continuation Reactor Temperature vs Residence Time



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0.06 time [s]

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One parameter continuation Reactor Temperature vs Equivalence ratio



For fixed residence time, the change of reactor temperature respect to the inlet mixture composition.

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One parameter continuation Reactor Temperature vs Equivalence ratio



For fixed residence time, the change of reactor temperature respect to the inlet mixture composition.

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Multi-P Cont.

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One parameter continuation

Reactor Temperature for non-adiabtic reactors



Dependence of reactor temperature on residence time for non-adiabatic PSR $p = 1 \text{ atm}, T_0 = 700 \text{ K}, \phi = 1 \text{ and } \dot{Q}_{loss} = 0.1 kJ/(s \times m^3)$

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1P Cont.

T vs ø

T vs Q_{lor}

Multi-P Con

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Multi-parameter continuation Continuation in $(T_0 - \tau)$ parameters



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Multi-parameter continuation Continuation in $(T_0 - \tau - \phi)$ parameters

1.2 1.1 1 ϕ 0.9 0.8 0.7 $10^{-6} \ 10^{-5} \ 10^{-4} \ 10^{-3} \ 10^{-2} \ 10^{-1}$ 10^{0} 10^1

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Directions for future work

Directions for future work

• Entropy Production Analysis Direct Numerical Simulations using skeletal mechanisms CH₄/air premixed flame (p = 1 atm, $\phi = 0.9$, $T_0 = 300$ K, $\delta_f = \frac{T_f - T_0}{max^2 (dT_1)}$)



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Directions for future work

Entropy Production Analysis

Direct Numerical Simulations using skeletal mechanisms CH₄/air premixed flame (p = 1 atm, $\phi = 0.9$, $T_0 = 300$ K, $\delta_f = \frac{T_f - T_0}{max|\frac{dT}{dx}|}$)



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Entropy Production Analysis

Direct Numerical Simulations using skeletal mechanisms

CH₄/air premixed flame (p = 1 atm, $\phi = 0.9$, $T_0 = 300$ K, $\delta_f = \frac{T_f - T_0}{max|\frac{dT}{dt}|}$)



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Spectral Quasi Equilibrium Manifold for Chemical Kinetics. In preparation for

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